

AIR TIME AND

ENGINEERING RECORD			
DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFG.
	HRS.	MIN.	HRS. MIN.
BROUGHT FORWARD			1:50 2

NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETION.

01.04	04	0	.8	1050	2
02.12	04	0	.6	1050	3
04.24	04	0	.5	1051	0.3
04.24	04	0	.4	1051	3
04.28	04	0	.5	1052	2
04.30	04	0	.5	1052	3
05.04	04	0	.5	1053	2
05.04	04	0	.6	1053	8
05.05	04	0	.3	1054	1
05.07	04	0	.7	1054	9
05.16	04	0	.5	1055	3
05.19	04	0	.5	1055	8
05.28	04	0	.5	1056	3
05.29	04	0	.5	1056	8
05.29	04	0	.5	1057	3
05.29	04	0	.3	1057	6
06.10	04	0	.9	1058	—
06.12	04	0	.6	1058	6

06	19	04	1	1	1059	7
06	19	04	0	8	1060	5
06	23	04	0	5	1061	0
07	03	04	0	7	1061	7
07	03	04	0	6	1062	3
07	03	04	0	8	1063	1

CARRIED FORWARD

ENGINEERING RECORD

INSPECTION, REPAIRS & REPLACEMENT

ENTRIES. DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS

April 23, 2004
Engine removed and dismantled by owner after detection of crack in engine case. Case overhauled by Divco-wind 74038. Engine reassembled using new main bearings and gaskets IAW Lymcom overhaul manual section 7. Engine re-installed IAW Commander mm. SEC 4-24 8L Phillips 20W50 oil added. Engine ground run-temps and pressures normal

entered out of sequence

June 9, 2004
Oil pressure relief ball and spring replaced with new. 4 AN960 washers added. Oil pressure within normal range on ground run-up. This maintenance release is contingent on a satisfactory test flight to ensure acceptable cruise oil pressure.

THE MAINTENANCE DESCRIBED ABOVE
HAS BEEN PERFORMED IN ACCORDANCE
WITH THE APPLICABLE STANDARDS
OF AIR WORTHINESS.

PA744252
PA744251
PA744250
PA744249
PA744248

AIR TIME AND

[illegible]

ENGINEERING RECORD

[illegible]

August 26, 2004

Left foot step removed and reinstalled after weld repair
by Brant Aero w/o 0410597

THE MAINTENANCE DESCRIBED ABOVE
HAS BEEN PERFORMED IN ACCORDANCE
WITH THE APPLICABLE STANDARDS
OF AIR WORTHINESS.
AIRCRAFT NO. 7445

47004

REMARKS:

AIR TIME AND

[illegible]

CARRIED FORWARD

ENGINEERING RECORD

INSPECTION, REPAIRS & REPLACEMENT

ENTRIES. DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

SIGNATURE

October 18, 2001
 104th Annual in person complete! AW C&E £25 asp. B. 21, ind fill & bring id 6L
 Philip 20w56 id adient. Cars Test 1 177, 272, 63 75, 44 77. Flg. ink to p.
 bolt re-touched. Touch check complete!
 Compass awning N 30 60 90 120 150 180 210 240 270 300 330
 01 32 62 81 121 152 132 211 241 271 307 337
 AD 90-03/2 AD 76-07-12 completed. Engine ground nut - temp. id on brass
 normal

ELT RECERTIFIED BY DISCOUNT MONTH 42010

THE MAINTENANCE LOGS
HAS BEEN PERFORMED IN ACCORDANCE
WITH THE APPLICABLE STANDARDS
OF AIR WORTHINESS
AIRMObILE W/O# 2438

AIR TIME AND

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFG.	
	HRS.	MIN.	HRS.	MIN.
BROUGHT FORWARD			1083	9
10 23 04	02	00	1085	9
10 28 04	0	6	1086	5
11 03 04	1	0	1087	5
11 07 04	1	7	1089	2
11 12 04	1	5	1090	7
12 04 04	0	4	1091	1
12 17 04	0	6	1091	7
12 27 04	0	6	1092	3
01 14 05	0	6	1092	7
01 25 05	0	6	1093	5
02 01 05	0	5	1094	0
02 11 05	0	4	1094	4
02 13 05	0	8	1095	2
02 26 05	0	6	1095	8
03 04 05	0	8	1096	6
03 17 05	0	5	1097	1
03 26 05	0	5	1097	6
04 06 05	1	1	1098	7
04 08 05	0	7	1099	4
04 09 05	0	8	1100	2
04 13 05	0	7	1100	9
CARRIED FORWARD				

Mar. 04 2005 - C-FDJJ - Repair loop antenna connector wiring, and KI225 indicator wiring. Reset transponder pins. Ramp tested ok. The maintenance described has been performed in accordance with applicable airworthiness requirements. Further details on Brant Aero work order #05-15823 A.M.O. #10-74

Pubh

ENGINEERING RECORD

10

INSPECTION, REPAIRS & REPLACEMENT		SIGNATURE	LICENCE NUMBER
ENTRIES: DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.			
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AIR TIME AND

ENGINEERING RECORD				
DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFD.	NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETION
	HRS.	MIN.	HRS. MIN.	
BROUGHT FORWARD	1100	54	1100 9	
04 14 05	0	5	1101 4	
04 15 05	0	8	1102 2	
04 16 05	0	8	1103 0	
04 18 05	0	9	1103 9	
04 19 05	2	9	1106 8	
04 21 05	3	1	1109 9	
04 22 05	1	0	1110 9	Compass no longer accurate
05 18 05	0	6	1117 53	
05 20 05	0	4	1111 9	
05 20 05	0	4	1112 3	
05 23 05	0	4	1112 7	
06 10 05	0	4	1113 1	
06 11 05	0	8	1113 9	
06 20 05	1	3	1115 0	
06 24 05	0	6	1115 6	
07 01 05	0	4	1116 0	
07 19 05	0	6	1116 6	
07 19 05	0	5	1117 1	
07 25 05	0	9	1118 0	
07 29 05	0	4	1118 4	
07 30 05	2	0	1120 4	
07 30 05	2	9	1123 3	
08 03 05	2	1	1125 4	

CARRIED FORWARD

ENGINEERING RECORD

[illegible]

1. COUNTRY
UNITED
KINGDOM

2 Civil Aviation Authority
A member of the JAA

technical data Copies: Green - Consignor, White - Consignee, Yellow - Consignee

3. CERTIFICATE REF. NO.
SIRS/73852/33

JAA FORM ONE

AUTHORISED RELEASE CERTIFICATE

4. ORGANISATION

S.I.R.S. Navigation Ltd.,

Compass House, Bowes Estate, Wrotham Road, Meopham, DA13 0QB, England



5. WORK ORDER/CONTRACT
16157

6. ITEM

7. DESCRIPTION

8. PART No

9. ELIGIBILITY*

10. QTY.

11. SERIAL/BATCH No.

12. STATUS/WORK

1

Navigator Compass (12V BLUE
ILLUMINATION)

NV2C - 12V

Various

1

Ser: 621021-33

Manufactured

Approved to TSO C7d
Ref:GE/vk/11/18/98/C7d
And CAA Approval E15753

B/No. 621021

13. REMARKS

Manufactured in accordance with specification NV5125 Issue 2
This Certificate has been issued under National Rule Provisions

Limited life parts will normally be accompanied by maintenance history including life used

14. ☒ Airworthiness ☐ Conformity only

Certifies that the part(s) identified above except as otherwise specified in block 13 was (were)
manufactured/inspected in accordance with the applicable design data and with the Airworthiness regulations of
the stated country (See below)

19. ☐ JAR-145.50 Release to service

Certifies that the work specified above except as otherwise specified in block 13 was carried out in
accordance with JAR-145 and in respect to that work, the part(s) is (are) considered ready for release to service.
(see below)

15. Signature:

16. Approval Reference Number:
DA1/8992/85

20. Signature

21. Approval Reference Number:
CAA00413

17. Name: P. B. MARINER

18. Date (d/m/y): 17 March 2005

22. Name

23. Date (d/m/y)

USER/INSTALLER RESPONSIBILITIES

- It is important to understand that the existence of the document alone does not automatically constitute authority to install the part/component/assembly
- Where the user/installer works in accordance with the national regulations of an Airworthiness Authority different from the Airworthiness Authority specified in block 2 it is essential that the user/installer ensures that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority specified in block 2.
- Statements 14 and 19 do not constitute installation certification. In all cases the aircraft maintenance record must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.

JAA Form One Issue:3 * installer must Cross-check eligibility with applicable technical data

Kind PA 744282
PA 744282

AIR TIME AND

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFG.		ENGINEERING RECORD
	HRS.	MIN.	HRS.	MIN.	
BROUGHT FORWARD			1125	4	
08 03 05	3	8	1129	2	
08 13 05	0	5	1129	7	
08 13 05	0	5	1130	2	
08 15 05	0	5	1130	7	
08 15 05	0	5	1131	2	
08 24 05	0	8	1132	0	
08 25 05	0	8	1132	8	
08 25 05	0	8	1133	6	
09 03 05	0	6	1134	2	
09 08 05	0	8	1135	-	
09 09 05	2	6	1137	6	
09 13 05	1	0	1138	6	
09 14 05	1	4	1140	-	
09 24 05	1	8	1141	8	
09 28 05	0	4	1142	2	
09 30 05	3	0	1145	2	
10 02 05	1	0	1146	2	
10 28 05	0	6	1146	8	
10 29 05	1	7	1148	5	
11 04 05	2	3	1151	8	

CARRIED FORWARD

ENGINEERING RECORD

INSPECTION, REPAIRS & REPLACEMENT
ENTRIES: DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

SIGNATURE

Sept 02, 2005
Pilot side master cylinder o rings replaced. Brake bled IAW Commander M.M. sec6-20

THE MAINTENANCE DESCRIBED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH APPLICABLE AIR WORTHINESS REQUIREMENTS.
EAGLE ROCK
AVIATION WO# 1004

Sept 12, 2005

Vacuum pump replaced with overhauled unit 215cc s/n 09966. Engine run up temps and pressures normal

THE MAINTENANCE DESCRIBED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH APPLICABLE AIR WORTHINESS REQUIREMENTS.
EAGLE ROCK
AVIATION WO# 1013

Oct 26, 2005

100hr/annual insp completed IAW CAR 625 appB&C. Oil and filter changed 7L Shell 15w50 oil added. Comp test all 79/80. Left tire rotated STC SA00865SE Vortex generators installed. Seat belts repaired by Aircraft belts. Alt. enc xponder recertified by Brant Aero w/o 05-16533 Pilot-static system tested IAW CAR 625app C. ELT recertified by Discount w/o 52072. Tach stroboscopically tested. Engine ground run-temps and pressures normal

THE MAINTENANCE DESCRIBED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH APPLICABLE AIR WORTHINESS REQUIREMENTS.
EAGLE ROCK
AVIATION WO# 1045

1. Approving National Aviation Authority/Country: **FAA/UNITED STATES**

2. **AUT**

4. Organization Name and Address: **Aircraft Belts, Inc. 2000 Anders Lane Kemah, Tx.**

6. Item: 7. Description: 8.

1.	Standard Lap Belt.	1108
2.	Standard Lap Belt	5005
3.	1Pt Inertia Reel	Non
4.	1Pt Inertia Reel	Non

13. Remarks:
Details of this repair are on file referencing work order #
Repair Station YB1R632K
Line Item 1, 2, 3, & 4 Renovated Per Specification AB

14. Certifies the items identified above were manufactured in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in block 12. and described in block 13. was performed in accordance with CAR 571.10.

☐ Approved design data and are in a condition for safe operation

☐ Non-approved design data specified in block 13.

15. Authorized Signature:

16. Name (Typed or Printed):

1. Approving National Aviation Authority/Country		2. AUTHORIZED RELEASE CERTIFICATE		3. Form tracking No.	
Transport Canada		TCCA 24-0078		47-00	
4. Organization Name and Address		5. Work Order/Contract/Invoice		6. Status/Work	
Discount Avionics 1 John James Dr. Angus, ON L0M 1B3		Division of Laetrd Inc.		52072	
7. Description		8. Part Number	9. Eligibility	10. Qty.	11. Serial/batch #
AMERIKING AK-450		N/A	N/A	1	454053
12. Status/Work		13. Remarks			
"INSPECTED"		"Performance Test"			
ELT Certified IAW CAR 571 Appendix "G"					
Batt Due: OCT,2006 ELT Due: OCT,2006					
19. ☒ CAR 571.10 Maintenance Release.					
☐ Other regulations specified in block 13.					
Certifies that, except where otherwise specified in block 13, the work identified in block 12. and described in block 13. was performed in accordance with CAR 571.					
16. Certificate/Approval ref. No.		20. Authorized signature		21. Certificate/Approval ref. No.	
N/A				AMO # 47-00	
18. Date		22. Name		23. Date	
N/A		Carl Bertrand M396639		10/19/2005	

User/Installer Responsibilities

It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.

the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in block 12. and described in block 13. was performed in accordance with CAR 571.

installation certification. In all cases, aircraft maintenance records must include installation certification issued in accordance with the

1. Approving National Aviation Authority/Country: FAA/UNITED STATES		3. Form Tracking Number: 66526	
4. Organization Name and Address: Aircraft Belts, Inc. 2000 Anders Lane Kemah, Tx. 77565		5. Work Order/Contract/Invoice Number: 66526	
AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG			
6. Item:	7. Description:	9. Eligibility: ^a	10. Quantity:
1. Standard Lap Belt.		TSO Article N/A	2
2. Standard Lap Belt.		TSO Article N/A	2
3. 1Pt Inertia Reel		TBD by Installer	2
4. 1Pt Inertia Reel		TBD by Installer	2
8. Part Number:		11. Serial/Batch Number:	
1108203		10/05	
500574		10/05	
None		10/05	
None		10/05	
12. Status/Work:			
Repaired			
Repaired			
Repaired			
Repaired			
13. Remarks: Details of this repair are on file referencing work order # 66526 Repair Station YB1R632K Line Item 1, 2, 3, & 4 Renovated Per Specification ABIRPS1			
14. Certifies the items identified above were manufactured in conformity to: ☐ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 13.			
19. ☒ 14 CFR 43.9 Return to Service ☐ Other regulation specified in Block 13 Certifies that unless otherwise specified in block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.			
15. Authorized Signature:		21. Approval/Certificate No.: YB1R632K	
16. Approval/Authorization No.:		23. Date (m/d/y): 10/19/05	
17. Name (Typed or Printed):		22. Name (Typed or Printed): Marilyn Hoang	
18. Date (m/d/y):			
User/Installer Responsibilities			
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly.			
Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1.			
Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.			

FAA Form 8130-3 (6-01)

^ainstaller must cross-check eligibility with applicable technical data.

NSN: 0052-00-012-9005

Changed to 05-16533 SE Vortex generators installed. Seat belts rotated STC SA00865SE Vortex generators installed. Seat belts repaired by Aircraft Belts, Inc. Alt, enc xponder recertified by Brant Aero w/o 05-16533 Pilot-static system tested IAW CAR 625app.C. ELT recertified by Discount w/o 52072. Tach stroboscopically tested. Engine ground run-temps and pressures normal

REQUIREMENTS.
EAGLEROCK
AVIATION WO#

104548m19-1373
Snd 107 PA744282
Snd 107 PA744282
Snd 107 PA744282

Supplemental Type Certificate

Number SA00865SE

This certificate, issued to
Micro Aerodynamics, Inc.
4000 Airport Rd. Suite D
Anacortes, WA 98221 98221

certifies that the change in the type design for the following product with the limitations and conditions
therefor as specified herein meets the airworthiness requirements of Part 23 of the Federal Aviation

Regulations:

Original Product—Type Certificate Number: A12SO
Make: Commander Aircraft Company

Model: 112, 112TC, 112TCA, 114, 114A, 114B, 114TC

Description of the Type Design Change: Installation of vortex generators on the wings, horizontal tail surfaces
and vertical fin in accordance with Micro Aerodynamics Drawing Package MA2128, Revision IR, dated
November 1, 2000, and Installation Manual MA2129, Revision IR, dated December 7, 2000, or later FAA
approved revisions.

NOTE: If more than five vortex generators are damaged or missing, the aircraft is not airworthy.

Limitations and Conditions: Approval of this change in type design applies to the above model aircraft only.

This approval should not be extended to aircraft of these models on which other previously approved
modifications are incorporated unless it is determined that the relationship between this change and any of those
other previously approved modifications, including changes in type design, will introduce no adverse effect upon
the airworthiness of that aircraft. A copy of this Certificate must be maintained as part of the permanent records
for the modified aircraft.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the
other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until sus-
pended, suspended, revoked, or a termination date is otherwise established by the Administrator of the
Federal Aviation Administration.

Date of application: November 11, 2000
Date of issuance: January 9, 2001



Date received: _____
Signature of the Administrator: _____
(Signature)
Acting Manager, Seattle Aircraft
Certification Office
(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.
This certificate may be transferred in accordance with FAR 21.47.
FAA Form 8130-2 (10-69)

Changes to STC SA00865SE Vortex generators installed. Seat belts
repaired by Aircraft belts. Alt, enc xponder recertified by Brant Aero
w/o 05-16533 Pilot-static system tested IAW CAR 625app.C. ELT
recertified by Discount w/o 52072. Tach strobe

REQUIREMENTS.
EAGLEROCK

2. Owner - Propriétaire	Name - Nom	Registration Markings - Marquages d'immatriculation
	COBO TESTER	112A C-EDJJ

3. Type of Work - Type de Travail	Modification	Repair - Réparation
☒	☐	

4. Name and address of person or organization who accomplished the repair or modification
 Nom et adresse de la personne ou de l'organisme qui a effectué la réparation ou la modification
 EAGLE ROCK AVIATION
 NIAGARA DISTRICT AIRPORT
 N.O.T.L. LOS 150

5. Description of work accomplished - Description des travaux effectués
 SITC # SAGE 865 SE

- 1) INSTALLED MICRO VORTEX GENERATOR KIT
 1AW MICRO DYNAMICS INSTALLATION
 MANUAL, DOCUMENT MA2129 REV. 1R
 DATED 12/7/2000.
- 2) INSTALLED MINIMUM EQUIPMENT PLACARD
 ON INST. PANEL
- 3) WEIGHT & BALANCE CHANGE NEGLIGIBLE.

CONTINUING AIRWORTHINESS - IF AIRCRAFT
 REPAIRED - VORTEX GENERATORS MUST BE
 REFINED.

Signature of person submitting the report - Signature de la personne soumettant le rapport

Date of completion - Date d'achèvement

AIR TIME AND

DAILY TOTAL		TOTAL AIR TIME SINCE MFG.		NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETE
HRS.	MIN.	HRS.	MIN.	
		1151	8	

CFDJJ Rockwell 112 s/n 273 Work Report # 14138 January 2, 2006

A MX170 nav/com, a KI201C indicator and direction gyro were removed and replaced with a Garmin GNS430 nav/com/GPS and Century NSD360A HSI. A GA56 GPS antenna for the GNS430 was installed on the top of the aircraft with hardware supplied by the manufacturer. A KMA20 audio panel was replaced with a Garmin GMA340. A Narco AT50A transponder was removed and replaced by a Garmin GTX327. An Ameriking encoder was removed from the aircraft, and replaced with a Trans-Cal SSD30A-RS232 blind encoder. The encoder was mounted below the glove box with AN hardware. The existing attitude indicator was removed and replaced with a Century Attitude indicator. A Century roll servo was removed. A Century 2000 model AK1056 autopilot system was installed according to STC SA8534SW-D with kit 2K-2019. Avionics interfaced according to manufacturer's recommendations. Altimeter replaced with certified unit. Replaced battery cables. Replace filter for gyro vacuum system.

The Transponder/encoder system was tested and the correspondence to the primary altimeter done to 20,000 ft, according to CAR's 571, Appendix F. Static leak check carried out. GNS430 GPS Interference test carried out successfully.

Installation done with reference the following approved data:

STC SA04-97

STC SA8534SW-D

Installation done with reference to the following specified data:

- AC 43.13-1B, Ch 4, Sect 4 (Metal work)
- AC 43.13-1B, Ch 7, Sect 1 - 5, 6, 7, 11) (Hardware)
- AC 43.13-1B, Ch 11, Sect 4.5 - 9, 10 - 13, 14, 15, 16, 17 (Circuit breakers, wire, harnesses, terminals, grounding/Bonding/Marking & connectors)
- AC 43.13-2A, Ch 2 & 3 (Radio/Antenna installations)
- AC 43.13-2A, Ch 1, Para. 2 & 4-12 (Static Loads/General)
- Garmin GNS430 Installation manual 190-00140-02 Rev Q
- Garmin GTX327 Installation manual 190-00187-02 Rev K
- Garmin GA66 Installation manual 190-00094-00 Rev E
- Trans-Cal SSD120-30A-RS232 Installation manual 93005 Rev G
- Century NSD360A Bulletin 734 Rev 8 dated 08/15/00
- Century 2000 Bulletin 2123 dated 03/01/00 Rev 1 dated 03/13/03

Compatibility of this modification with previously installed equipment has been determined. Electrical load determined not to exceed system capacity.

The following documents are active and applicable:

NSD360A:

- Instruction for continued airworthiness dated 12/2/05
- Century NSD360A Bulletin 734 Rev 8 dated 08/15/00
- Century NSD360A Pilot Operating handbook 68S85

Garmin GNS430:

- STC SA04-97
- Master Drawing List MDL04030
- Flight Manual Supplement FMS04030-1
- Engineering Order EO04030-1
- Maintenance Manual Supplement MMS04030-1
- Drawing WDD04005

Century 2000 Autopilot:

- Century STC SA8534SW-D
- Century Bulletin # 2123 dated 03/01/00 Rev 1 dated 03/13/03
- Century Master Drawing list 87A1280 dated 03/01/00 Rev A dated 03/13/03 (included in Bulletin # 2123)
- Century Approved flight manual 68S1122 dated 03/30/00 Rev 1 dated 03/13/03 (included in Bulletin # 2123)
- Century Instruction for continuing Airworthiness 68S1140 Rev F dated 04/19/04
- Pilot Operating handbook 68S1035

No unique maintenance or inspection requirements are imposed by incorporation of these modifications. Subject to satisfactory test flight.

Items deferred:

- Pilot control column forward attaching point worn beyond limit.
- Weight and balance to be amended

TRANSCRIBED FROM
JOURNEY LOG

1151 8

CARRIED FORWARD

(REASON FOR REMOVAL)
(DETAILS OF WORK PERFORMED)
(ADDITIONAL WORK TO BE PERFORMED)
(INSTALLED ON)

WARRANTY VALIDATION

21	QC11	QC111	QC11V/ED	QC2000
Avionics, LTD				
ACFT DATA				
Date of inst.	112106			
Aircraft Model	112			
Aircraft S/N	273			
Reg. No.	CFD55			
Aircraft Hours	1151.8			
New/Used/Demo				

LEGEND:
1st Copy (White)
2nd Copy (Yellow)

03-01-00 Rev. A dated 03-01-00 Rev. 1 dated 03-01-00 Rev. A dated 03-13-03, (14 Volt Electrical System) or later approved Supplement to Pilot's Operating Handbook and/or FAA/DAS N 68S1122, 03-30-00 Rev 1 dated 03-13-03.

14A.
approved modifications must be determined by the installer.
this certificate to alter the product, the holder shall give the other person

which is the basis for approval shall remain in effect until
ation date is otherwise established by the Administrator of the

TORONTO AVIONICS LTD. PETERBOROUGH MUNICIPAL AIRPORT

NOMENCLATURE: 1451
MANUFACTURER: CEMURV TYPE/MODEL: NSD 360A
PART NO: 520136-1333 SERIAL NO: 15893A
TOTAL TIME: WORK ORDER: 14138
OVERHAUL: ☐ REPAIR: ☐ FUNCTIONAL CHECK: ☐ REALIGNED: ☐
REMOVAL: ☐ NEW: ☒ MODIFICATION: ☐
PREVIOUS CERTIFICATION:

The maintenance described has been performed in accordance with the applicable standards of airworthiness.

112106 Date
1122208 Lic No/Stamp

TORONTO AVIONICS LTD. PETERBOROUGH MUNICIPAL AIRPORT

NOMENCLATURE: ALTIMETER
MANUFACTURER: AEROSONIC TYPE/MODEL:
PART NO: 101735-01495 SERIAL NO: C 102598
TOTAL TIME: WORK ORDER: 14138
OVERHAUL: ☐ REPAIR: ☐ FUNCTIONAL CHECK: ☒ REALIGNED: ☐
REMOVAL: ☐ NEW: ☐ MODIFICATION: ☐
PREVIOUS CERTIFICATION: 36051

The maintenance described has been performed in accordance with the applicable standards of airworthiness.

112106 Date
1122208 Lic No/Stamp

TORONTO AVIONICS LTD. PETERBOROUGH MUNICIPAL AIRPORT

NOMENCLATURE: PITCH SERVO
MANUFACTURER: TYPE/MODEL:
PART NO: 15784-3-1056-8 SERIAL NO: 1491
TOTAL TIME: WORK ORDER: 14138
OVERHAUL: ☐ REPAIR: ☐ FUNCTIONAL CHECK: ☐ REALIGNED: ☐
REMOVAL: ☐ NEW: ☒ MODIFICATION: ☐
PREVIOUS CERTIFICATION:

The maintenance described has been performed in accordance with the applicable standards of airworthiness.

112106 Date
1122208 Lic No/Stamp

TORONTO AVIONICS LTD. PETERBOROUGH MUNICIPAL AIRPORT

NOMENCLATURE: TRANSDUCER
MANUFACTURER: GARMIN TYPE/MODEL: GTX 327
PART NO: SERIAL NO: 83722198
TOTAL TIME: WORK ORDER: 14138
OVERHAUL: ☐ REPAIR: ☐ FUNCTIONAL CHECK: ☐ REALIGNED: ☐
REMOVAL: ☐ NEW: ☒ MODIFICATION: ☐
PREVIOUS CERTIFICATION:

The maintenance described has been performed in accordance with the applicable standards of airworthiness.

112106 Date
1122208 Lic No/Stamp

subject to satisfactory test flight.
terms deferred:
Pilot control column forward attaching point worn
Weight and balance to be amended

REMARKS

DATE:

WORK TO BE PERFORMED UPON INSTALLATION

WORK PERFORMED

DATE:

WORK TO BE PERFORMED UPON INSTALLATION

LEASE

RENTON MUNICIPAL AIRPORT
RENTON, WASHINGTON 98115
FAA FORM 8130-2 (10-68)

CENTURY FLIGHT SYSTEMS, INC.	
11250 11TH AVE. S.W. ALBUQUERQUE, N.M. 87105	
MFR	AERO COMMANDER
MODEL	112
ACT. SIN.	273
REG. NO.	GFDJJ
GFS SIN.	2K-2019
ORD. NO.	65016
ONLY CORPORATE SEALED SET COPY IS VALID	

Number SA0534SW-D

Century Flight Systems, Inc.
3003 FM 1195
P.O. Box 610
Mineral Wells, TX 76067

This certificate issued to
certifies that the change in type design for the following product with the
therefore as specified herein meets the airworthiness requirements of Part 23 of the Federal Aviation
Regulations.

Original Product — Type Certificate Number: A12SO

Make: COMMANDER

Model: 112, 112B, 112TC, 114 and 114A

Description of Type Design Change: Installation of a Century Automatic Flight System, Model AK1056 consisting of a Century 2010/2030/2031/2050/2051 Autopilot per Bulletin No. 2123 dated 03-01-00 Rev 1 dated 03-13-03 and Master Drawing List 87A1280, dated 03-01-00 Rev. A dated 03-13-03, (14 Volt Electrical System) or later FAA/DAS Approved Revision. FAA/DAS Approved Supplement to Pilot's Operating Handbook and/or FAA/DAS Approved Airplane Flight Manual Supplement P/N 68S1122, 03-30-00 Rev 1 dated 03-13-03.

Limitations and Conditions:

Eligible on Commander 112, 112B, 112TC, 114 and 114A.

Compatibility of this modification with other previously approved modifications must be determined by the installer.

If the holder agrees to permit another person to use this certificate to alter the product, the holder shall give the other person written evidence of that permission.

This certificate and the supporting data which is the basis for approval shall remain in effect until
surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the
Federal Aviation Administration:

Date of application: 10-25-99

Date revised:

Date of issuance: 03-31-00

Date amended: 03-13-03 Rev 1

By direction of the Administrator



William R. Watson

(Signature)

William R. Watson

DAS Staff Coordinator DAS T SW

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.
FAA Form 8130-2 (10-68)

This certificate may be transferred in accordance with FAR 21.41

MAJOR REPAIR OR MAJOR MODIFICATION REPORT RAPPORT DE RÉPARATION MAJEURE OU DE MODIFICATION MAJEURE

INSTRUCTIONS

For instructions, see Standard 571 Appendix L - Pour les instructions, consulter l'appendice L à la norme 571

1. Aircraft - Aéronef	Make - Marque PERO COMMANDER Serial No. - N° de série 273	Model - Modèle 112
2. Owner - Propriétaire	Name - Nom GORD TESSIER	Registration Marks Marques d'immatriculation C-FDJJ
3. Type of Work - Type de Travail ☒ Modification ☐ Repair - Réparation		
4. Name and address of person or organization who accomplished the repair or modification Nom et adresse de la personne ou de l'organisme qui a effectué la réparation ou la modification EAGLEROCK AVIATION NIAGARA DISTRICT AIRPORT VIRGIL, ONT. L0S 1T0		

5. Description of work accomplished - Description des travaux exécutés

STC SA431SW

Work Report

Remove existing voltage regulator and overvoltage sensor.
 Installed new R15V0N Alternator control unit IAW Zeftronics drawing list, dated 9-13-91.

No weight and balance change. Equipment list updated
 Engine ground run, charging system operation normal

There were no published continuing airworthiness information provided with this STC.

Signature of person submitting the report - Signature de la personne soumettant le rapport

Date of completion - Date d'achèvement

MAR-2-06

THAT WERE AFFECTED BY THE WORK ACCOMPLISHED. **SA 431SW**

THE MAINTENANCE DESCRIBED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH THE REQUIREMENTS OF THE AIR WORTHINESS REGULATIONS. **112**

Mar 14, 2006
 Left main tire replaced with 600x6 6 ply Airhawk.
 Bearings cleaned and repacked.

United States of America
Department of Transportation — Federal Aviation Administration
Supplemental Type Certificate

Number **SA8431SW**

This certificate issued to **Tovya Group, Inc. dba Zeftronics**
1622 E. Whaley St.
Longview, TX 75601-6830

certifies that the change in the type design for the following product with the limitations and conditions
hereof as specified hereon meets the airworthiness requirements of Part * of the

Regulations:

Original Product — Type Certificate Number: * *See
Make: * attached FAA Approved Model List
Model: * No. SA8431SW for list of approved
aircraft and certification basis.

Description of Type Design Change:

Installation of R15VON Alternator Control Unit on aircraft shown
on Approved Model List, in accordance with Tovya Group, Inc. dba
Zeftronics, drawing list ZOFPCE, dated 9-13-91, or R1510N
Alternator Control Unit with V1510A Overvoltage Sensor in
accordance with drawing lists ZOAPCE, dated 5-13-91 and ZOBPCE,
dated 9-13-91, or later FAA approved data.

Limitations and Conditions:
Compatibility of this modification with previously installed
equipment must be determined by installer.

This certificate and the supporting data which is the basis for approval shall remain in effect until sur-
rendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the
Federal Aviation Administration.

Date of application: December 22, 1991

Date issued:

Date of issuance: August 26, 1992

Date amended: 01/06/94 Rev. 1



By direction of the Administrator

Mark R. Schilling (Signature)
Mark R. Schilling, Manager
Special Certification Office

(Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.
This certificate may be transferred in accordance with FAR 21.47.

144 44
THE MAINTENANCE DESCRIBED ABOVE
HAS BEEN PERFORMED IN ACCORDANCE
WITH THE MAINTENANCE INSTRUCTIONS
WHICH ARE AVAILABLE TO THE
REGULATORY AGENCY
FAA FORM 8110-2
144 44
PART WERE AFFECTED BY THE WORK
ACCOMPLISHED. See 2344282

Mar 14, 2006
Left main tire replaced with 600x6 6 ply Airhawk.
Bearings cleaned and repacked.

17. Name (Typed or Printed)
Femi G. Il
FAA Form 8130-3 (11-93)

Airworthiness Approval Tag

User/Installer Responsibilities

Where the user/installer work in accordance with the national regulations of an Airworthiness Authority different than the Airworthiness Authority of the country specified in block 1 it is essential that the user/installer ensures that his/her Airworthiness Authority accepts parts/components/assemblies from the Airworthiness Authority of the country specified in block.

The FAA Form 8130-3 and JAA Form One are equivalent. Other countries such as Canada also have equivalent acceptable documents.

The FAA Form 8130-3 and JAA Form One are equivalent. Other countries such as Canada also have equivalent acceptable documents.

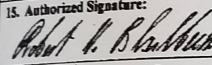
FAA Form 8130-3 (11-93)

Instructions

THAT WERE AFFECTED BY THE WORK

2A744282

Mar 14, 2006
Left main tire replaced with 600x6 6 ply Airhawk.
Bearings cleaned and repacked.

1. Approving National Aviation Authority/Country: FAA/United States		AUTHORIZED RELEASE CERTIFICATE FAA Form 8130-3, AIRWORTHINESS APPROVAL TAG				3. Form Tracking Number: RVB DAR LOG 01-15-06 AMERI-KING 5. Work Order/Contract/Invoice Number: B7793	
4. Organization Name and Address: AMERI-KING CORP. 17881 SAMFSON LANE HUNTINGTON BEACH, CA 92648							
6. Item:	7. Description:	8. Part Number:	9. Eligibility: *	10. Quantity:	11. Serial/Batch Number:	12. Status/Work:	
1	RFI FILTER P/N AK-950-FTR FAA PMA STC APPROVED	AK-950-FTR	SEE ATTACHED FAA APPROVED MODEL LIST (AML) MDL-950	2	S/Ns 484178, 483967	NEW	
NOTHING FOLLOWS							
13. Remarks: EXPORT - CANADA FAA PMA							
14. Certifies the items identified above were manufactured in conformity to: ☒ Approved design data and are in a condition for safe operation. ☐ Non-approved design data specified in Block 13.							
19. ☒ 44 CFR 43.9 Return to Service ☐ Other regulation specified in Block 13 Certifies that unless otherwise specified in Block 13, the work identified in Block 12 and described in Block 13 was accomplished in accordance with Title 14, Code of Federal Regulations, part 43 and in respect to that work, the items are approved for return to service.							
15. Authorized Signature: 		16. Approval/Authorization No.: DARF346015NM		20. Authorized Signature:		21. Approval/Certificate No.:	
17. Name (Typed or Printed): ROBERT V. BLACKBURN		18. Date (m/d/y): 01-15-06		22. Name (Typed or Printed):		23. Date (m/d/y):	
User/Installer Responsibilities							
It is important to understand that the existence of this document alone does not automatically constitute authority to install the part/component/assembly. Where the user/installer performs work in accordance with the national regulations of an airworthiness authority different than the airworthiness authority of the country specified in Block 1, it is essential that the user/installer ensures that his/her airworthiness authority accepts parts/components/assemblies from the airworthiness authority of the country specified in Block 1. Statements in Blocks 14 and 19 do not constitute installation certification. In all cases, aircraft maintenance records must contain an installation certification issued in accordance with the national regulations by the user/installer before the aircraft may be flown.							

Airtex PRODUCTS, INC.

259 Lower Morrisville Road Fallsington, PA 19054 • 215/295-4115 FAX: 215-295-4417

SOLD TO
• GORD TESSIER
4 GOODMAN CRESCENT
ST. CATHERINES, ONTARIO
CANADA L2S-3Y7

SHIP TO
STATE FARM
C/O GORD TESSIER
395 ONTARIO ST.
UNIT D2
ST. CATHERINES, ONTARIO
CANADA L2N-7M6

DATE ENTERED:

10/11/05

DATE SHIPPED:

10/12

INVOICE DATE:

VIA: UPS/COM

YOUR ORDER:

OUR ORDER:

33820

TERMS:

CREDIT CARD ☒

FALLSINGTON

F.O.B.:

QUANTITY	905 641 1819	DESCRIPTION	PRICE	AMOUNT
1		ROCKWELL COMMANDER 112A 1975		293.00
		CARPET SET- NAVY		32.55
		SHIPPING		325.55

I CERTIFY THAT THE GOODS REFERENCED IN THIS INVOICE COMPLY WITH THE ORIGIN REQUIREMENTS SPECIFIED FOR THESE GOODS IN THE NORTH AMERICAN FREE TRADE AGREEMENT, AND THAT FURTHER PROCESSING OR ASSEMBLY OUTSIDE THE TERRITORIES OF THE PARTIES HAS NOT OCCURRED SUBSEQUENT TO PROCESSING OR ASSEMBLY IN THE NAFTA REGION.

ALL OF THE ABOVE MATERIALS ARE CERTIFIED TO COMPLY TO FAR 23.853(a) HORIZONTAL BURN SPECIFICATIONS THAT ARE REQUIRED BY THE FAA AND DONE ACCORDING TO THE QUALITY CONTROL MANUAL DATED 10-12

WITH THIS ORDER PLEASE REFER TO OUR ORDER NUMBER.

ALL REPORT ABOVE ENGINE CONTROLS THAT WERE AFFECTED BY THE WORK ACCOMPLISHED.

PA344282

Mar 14, 2006
Left main tire replaced with 600x6 6 ply Airhawk.
Bearings cleaned and repacked.

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFG.	
	HRS.	MIN.	HRS.	MIN.

DATE	DAILY TOTAL		TOTAL TIME SINCE MFG.	
	HRS.	MIN.	HRS.	MIN.
BROUGHT FORWARD			1163	10
March 03/06	0	7	1163	.8
March 04/06	1	0	1164	.8
March 05/06	1	0	1165	.8
March 06/06	0	8	1166	6
March 07/06	1	0	1167	6
March 20/06	0	9	1168	5
March 21/06	1	7	1170	2
March 26/06	0	6	1170	8
March 27/06	1	5	1172	3
March 02/06	0	7	1173	0
April 02/06	0	8	1173	8
April 01/06				
April 11/06	0	8	1174	6
April 15/06	0	7	1175	3
April 18/06	0	0	1176	3
April 18/06	1	0	1177	3
April 18/06	0	3	1177	6
April 27/06	0	7	1178	3
April 29/06	0	9	1179	2
April 29/06	0	9	1180	1
May 2nd/06	0	4	1180	5
May 2nd/06	0	8	1191	3
May 19/06	1	0	1192	3
May 29/06	1	0	1193	4
June 01/06	1	0	1194	4

CARRIED FORWARD *entered in memo ⑧*

INSPECTION, REPAIRS & REPLACEMENT

ENTRIES. DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

April 7, 2006
Aircraft weighed IAW Commander Aircraft weight up
procedure. New weight and balance calculated. Oil and filtre
changed, 6L Shell 15W50 oil added. New fuel sump drains p/n
CAV170 installed. Engine ground run-temps and pressure
normal

THE MAINTENANCE DESCRIBED ABOVE
HAS BEEN PERFORMED IN ACCORDANCE
WITH APPLICABLE AIR WORTHINESS
REQUIREMENTS.

~~FROM~~

ENGINEER
AVIATION WO#

1135

FOR	N	30	60	E	120	150
STEER	N	30	60	90	120	150
FOR	S	210	240	W	300	330
STEER	180	210	240	270	300	330
DATE	MAY - 1 - 66 SIGN <i>[Signature]</i>					
RADIO ON	☒	OFF <i>[Signature]</i>				

Sd / PA744282
Sd / PA744282
Sd / PA744282
Sd / PA744282

AIR TIME AND

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFS.		ENGINEERING RECORD NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETE
	HRS.	MIN.	HRS.	MIN.	
BROUGHT FORWARD	1183	4			
June 01. 2006	1	0	1183	4	
June 02. 2006	0	9	1185	3	
June 03. 2006	1	0	1186	3	
June 06. 2006	1	0	1187	3	
June 07. 2006	1	0	1188	3	
June 09. 2006	1	3	1190	6	
June 10. 2006	1	0	1190	6	
June 13. 2006	1	0	1191	6	
June 14. 2006	1	0	1192	6	
June 15. 2006	1	3	1193	9	
June 17. 2006	1	3	1195	2	
June 23. 2006	1	3	1196	5	
June 25. 2006	0	9	1197	4	
June 25. 2006	0	9	1198	3	
June 30. 2006	1	3	1199	6	
July 01. 2006	1	3	1200	9	
July 04. 2006	1	5	1202	4	
July 05. 2006	0	5	1202	9	
July 06. 2006	1	1	1204	-	
July 07. 2006	1	0	1205	-	
July 08. 2006	1	0	1206	-	
July 13. 2006	1	4	1207	4	
July 17. 2006	3	2	1210	6	
July 18. 2006	1	0	1211	6	
July 19. 2006	1	0	1212	6	
July 25. 2006	0	8	1213	4	
July 26. 2006	0	3	1213	7	
July 26. 2006	0	5	1214	2	
July 29. 2006	1	0	1215	2	
Aug 17/06	5	1	1220	3	
11/18/06	9	4	1228	7	CARRIED FORWARD

ENGINEERING RECORD

17

INSPECTION, REPAIRS & REPLACEMENT
ENTRIES: DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

SIGNATURE

LICENCE NUMBER

June 22. 2006
Prop removed and re-installed top facilitate replacement of crank seal and alternator belt. #1 cylinder baffle patched.
Engine washed

THE AIRCRAFT HAS BEEN PERFORMED IN ACCORDANCE WITH APPLICABLE AIR WORTHINESS REQUIREMENTS.
EAGLEROCK
AVIATION WFO

July 5. 2006

Alt. belt retensioned. Oil pres. checked with test gauge. Found to be within limits.

MAINTENANCE DESCRIBED ABOVE HAS BEEN PERFORMED IN ACCORDANCE WITH APPLICABLE AIR WORTHINESS REQUIREMENTS.
EAGLEROCK
AVIATION WFO

1206

1197

1206

1206

1206

1206

1206

1206

1206

1206

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1206

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1206

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1206

1206

AIR TIME AND

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE WFO.	
	HRS.	MIN.	HRS.	MIN.

NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETE

BROUGHT FORWARD - - 1218 7

Aug 19/06 1 0 1219 7

Aug 20/06 1 0 1230 7

WORK CHW PER HRA WFO
FUEL LOG (15/50), AND
ON BOTH PLUGS. STARTS -
CHANGED THIS DATE, MAINT
(COPY ATTACHED TO WFO)

Aug 24/06 2 3 1232 9

Aug 23/06 9 1233 8

Nov. 24/06 - - 1233 8

FOUNDING IS FOR REF. RECORDS
CHW PER INSPECTION SHEETS
THE AIRCRAFT'S MAINT. RECORDS
ARE CONTAINED IN THOSE
ALL PROP. WORK I.A.D. AND
RESPECTIVE LOGS - ALL AIRPLANE
PHASE TASKS LISTED IN
+ CORROSION CRANKED UP -
ROTOR WING FROM EXHAUST -
WAKE IN BATTERY BOX + SERVICED
REMOVED, DISASSEMBLED, ALL CORROSION
PUMP AND BRASS FITTING FRAMES
FOR SPILL LINE SERVANT (OPS.)
WH TANKS FUEL TO CHECK
SET SCREW IN CONTROL KNOB
- AIRCRAFT JACKED, ALL (3) CNG
M₂ TO SPEC, AIR "WHITE"
CYLINDER O-RINGS REPLACED,
SECC, ALL BRAKE PADS RE-
PROPERLY - REPAIRED WH CABIN
SWITCH - SHUTTY DUTY RE
WITH MIL-H-5606 - EMERGENCY
LOCKED IN "NORMAL" POSN.
BELL CRANK BEARING HINGE BOWS
END OF ELEVATOR PUSH ROD @
ON WS, FILED WITH HRA WFO

1233 8

CARRIED FORWARD

ENGINEERING RECORD

18

INSPECTION, REPAIRS & REPLACEMENT
ENTRIES: DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

SIGNATURE
LICENCE NUMBER

#2607 - INFO AS FOLLOWS: OIL &
FILTER CAT OPEN - M₂ LEVEL SET
DUE TO WFO #2607 "MOT" BEING
REVERSE CHW IN JOURNAL COG
- ENTRY BY J.E. - M.E. FRAME M123370

PURPOSES ONLY. 100 HRA/ANNUAL INSPO.
WORK SHEETS, SUPPORTING LOGS FILED IN
WITH HRA WFO #2619 - ALL MAINT. REFUSES
DOCUMENTS - BRIEF DESCRIPTION IS FOLLOWS:
ALL ENGINE WORK/AD'S LISTED IN THEIR
AND MAINT. ACCESSORY AD'S + OUT OF
SECTION II LOGS - GASCHUTTER CRANKED
LANDING UNIT WIKES REPAIRED AND RE-
CLEANED UP CORROSION ON FITTINGS/HAND-
BATTERY - ALL FUEL DRAINED, FUEL SERVICED
CRANKED UP, "O"-RINGS REPLACED, S/S PISTON
TESTS NORMAL) R/H AND THEN ALTERNATE
FOR INTERNAL TRANSFER + LEAKS - REPAIR
+ REPAIR "OFF" POSN. LOCK OUT ON SELECTOR
STARTS RE-SERVICED WITH MIL-H-5606 +
BEARINGS RE-PACKED, BOTH BRAKE SQUE
BRAKES PULSED + BRED WITH FRESH MIL-H.
PLACED - BEACON/STROBE SWITCHES RE-INSTALLED
HRT CONTROL - REPAIRED PLUGS AND LIGHT
+ RE, ALL O-RINGS REPL. + UNIT SERVICED
GEAR EXTENSION KNOB REMOVED TO KEEP IT
UNTIL INTENTIONALLY SELECTED - ALTERN
1 MTS TIGHTENED - REPAIRED BOLT IN LOWER
BELL CRANK IN TAIL CONE (OIL INSPO. SIGNED
#2619 IN MAINT RECORDS) (CONT. NEXT PAGE - J.E.)

M.E. FRAME
M123370

AIR TIME AND

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE NFS.		NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETE
	HRS.	MIN.	HRS.	MIN.	
BROUGHT FORWARD	-	-	1333	8	(CONT. FROM PREVIOUS PAGE) ALL MAINT. AND WORK SHEETS FILED IN THE ALCON DEFERRED DEFERRERS ARE AS FOLLOWS: 1) PULL TUBE @ BELL CRANK IN THE DUTY 2) BEARINGS IN ALCON BELCRANKS TO ROCK UP & DOWN (POTENTIAL) 3) LARGER AND/OR FUEL TRANSFER OCCURS 4) ENGINE WILL REQUIRE FUEL TANK DRIPS FUEL OUT OF FITTINGS IN TANK??) ONLY ADDITIONAL WORK CLW THIS DATE VENT SYSTEM REPAIRED AND VENT LEAK - A.D. & CF-90 OR (12) AND OUT OF PHASE TASKS TO MATCH DEC 406 AS STAIN. RELEASE IS IN TOWARD LEFT WING TIP IN (ABOVE TRANSMISSION) FRONT
NOV. 30/67	-	-	1333	8	
DEC. 1/67			1333	8	
DEC. 4/67	9		1334	7	WORK CLW PER HRA WIA # 70 AS TO FIBROUS TIP WITH HIS PREVIOUSLY REPAIRED HUNGAR AND WING INSPECTED AS FOLLOWS: SPAR CRACKED (NO DAMAGE), WING ALCON POINT LOCATIONS BEARINGS - ALCON DISASSEMBLED FOR ENOUGH KINKS WORKED OUT (FROM PREVIOUS STOP DRILLED AND DRESSED (ALSO OF INTERNAL CORROSION) CLEANED STRIPPED CORROSION CLEANED UP & UP (CORROSION) AND BAL WEIGHT AND ALCON REINST. WITH AL HE MAY WANT TO REPLACE ALCON - FIBROUS REPAIRS CLW ON TIP - BIRD HOLE REPAIRED FROM LH WING -
DEC. 4/67	-	-	1334	7	
2007	-	-	1334	7	
JAN. 13/67	-	-	1334	7	
			1334	7	CARRIED FORWARD U.G.'s / AIRCRAFT STILL SER-

ENGINEER

ENTRIES. DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

SIGNATURE

LICENCE NUMBER

PERUSES/DUAL INSPECTIONS SIGNED ON INSP. RECORDS (THIS DATED) WITH HM W/ 3619 -

1) BUSHINGS IN LOWER END OF FUELSTER PUSH REQUIRE REPL. NEW PUMP (PUMP) (PUMP)

HAVE SAME SIDE PUMP AND ROAT ALONG WITH

2) FUELSTER FUEL SECUER FOR ANY EXT.

(GAS CORROSION DAMAGES REEL IF EITHER

(4) NEW PUMP NEXT ANNUAL, 2001

BEING NEXT WITH FUEL (CATER VIKES, DUE 22 IN 2 YRS. (SEE BROW)

- 4th FUEL TANK ALIGNED, OPENED UP

TANK RE. SEALED FUELED (NO LEAKS OR

AND 2002 - 2003 CFW AGAIN (ENGINE LOG)

ST GAS (2) (7), (7), + (10) CFW AGAIN

ANNUAL INSPECTION/REQUIRE DATE -

THIS JOURNAL LOG - ENTRY 27

GRASS ON LANDING (A FASTER SIGNATURE) (OWNER - PILOT)

JOURNAL LOG IS WRITTEN BY

3631 (W1) + W2) RISK OF DAMAGE ADDS MINOR DAMAGE TO ALUMINUM (NOTE: ALUMINUM RASH DAMAGE) - ALUMINUM + TIP RETAINED SPARS, STRAINERS, RAS AND REAR FUSEE ROOF AND GEAR WITH CHECKED (NO DAMAGE), PUSH RODS AND LINKAGES CHECKED (NO DAMAGE) TO ASSES NO FURTHER DAMAGE, STRAP BENDS HANGER RASH), (3) STRAP CLIPS TO TIE FOOT PREVIOUS DAMAGE), STRAP ATTACHED UP, BALANCE WEIGHT NOT RETURNED AND RE-POINTED, BAL. WEIGHT ANKLE NUTS CORRED REINST. WITH NEW BOLTS, HINGES REINST. NEW BOLTS/NUTS - (NOTE: OWNER ADVISED (OR ALUMINUM SKIN) FOR CERTIFICATES OTHERWISE) TIP RE-PAINTED + REINST. - REMAINS OF V.6. "SPARES" KIT CHECKED DUE TO (6) MISSING VISIBLE WITH MAX. (5) MISSING) (CONTINUED NEXT

N.E. FRAMER 111-23270

N.E. FRAMER 111-23270

AIR TIME AND

DATE	DAILY TOTAL		TOTAL AIR TIME SINCE MFS.		NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETE
	HRS.	MIN.	HRS.	MIN.	
BROUGHT FORWARD					
JAN. 11/02	-	-	12347		(CONT. FROM PREVIOUS PAGE) W/ TIME REMAINING W/ KIT ANNUAL WORK HAVE BEEN STARTED OUT
JAN. 26/02	-	-	12347		TO W/O 2631 - THIS ENTRY ADDITIONAL WORK C/W PER H/4 ON W/4 WING USING TUCO NEW
			12347		SERVICES R/C (3) C/O STAYS WITH PLUG (2) "O" RINGS, RE-SERVICES R/C ON W/O 2631 AND
			12347		CRASHED + FILED ON THIS
JULY 25/02	-	-	12347		THIS ENTRY FOR RECORDS FREE ONLY 2788 AND ITS ATTACHED
					FILED IN THIS RECORDS LOG.
					MENT OF STUDGAS (B)/(F/U) -
					OUT OF PHASE WORK RECORDED
					A.D.s AND OUT OF PHASE
					IN SECTION II LOG. ALL
					- REPAIRED EIT WITH NEW
					- INSTALLED "NEW" G35
					- REACHED ALL DRINGS IN
					- ALTIMETER SENT OUT FOR
					- RECHECKED THROTTLE ROD END
					DURING SEVEN REM.
					- RE + RE LOWER ELEVATOR
					BUT / BUSHING / SPACERS / SHIMS
					C/DUAL ON WORK
					- REPAIRED WINDSHIELD PER
					- CLEANED ALL GEAR SWITCHES
					- BRACES / PARK BRAKE
					- RE-ROUTED WIRING +
					TO ELIMINATE CONFLICT
					- LOWER COWL RE-
					NEW COMPASS CARD
					* ALSO SEE ADDENDUM /
					SHEETS FILED WITH W/O
					CARRIED FORWARD LOG + FRONT OF
					IN/ON WORK ORDER N/O ITS
			12347		

ENGINEERING RECORD

INSPECTION, REPAIRS & REPLACEMENT
ENTRIES. DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

SIGNATURE
LICENCE NUMBER

43631 HAS BEEN LEFT DOWN @ THIS
BUT MAINT. REPAIRS COVERING THIS
ON WORK SHEETS "WS1" & WS2 ATTACHED
M.E. FRAME M123370
AS FOLLOWS: REPAIRED (S) V.G.'s
V.103600, V.10360 & RECOVERED (1) - RE-
SHARPENED. RE-RE SHARPENED DANGER & RE-
M. - M.5206 - ALL FUNDAMENTAL REPAIRS
WITH ATTACHED WORK SHEETS WS1, 2 & 3
M.E. FRAME M123370
RECORDS - ENTRY AT PER
ALL WORK CARRIED OUT PER HRA W/L #
WORKING SHEETS & SUPPORT DOCS (ALL
ANNUAL ENDS. CLW TO FIN. REQUIRE-
ALL ENGINE / PROB. WORK, A.D., &
IN THEIR RESPECTIVE LOGS - ALL AIR
TAKES PER STD 625 (C) ARE RECORDED
OTHER WORK AS FOLLOWS: -
(AFTER CERT.) (SN-470839) -
BATTERY SN G03396019
SUPT. DRAINS, FUEL CAPS & GAS EXHAUST
CH (FOR STD 625 (C) (13)) -
+ NET. AIR LOCK CLEANS FIN -
(OVER SIGNED ON WORK SHEETS) -
ROD END IN FIN & RECONFIGURED
PROPERLY TO RETAIN FLEET PLY.
SHEETS)
AC 43-13.1B (3-40) (a +
+ SWUNG GEAR -
UNITS BLED.
REVERSED ROIT IN UN STD. BUNGEI
WITH MICRO SW. WIRING.
PAINTED -
HALDEX INST.
ALERT SHEET & DEFERRED WORK
2788 (AND) COPIES PLACED IN JOURNEY
THIS BUNGEI. ALL REQUESTS AND QUALS SIGNED
INST. WORK SHEETS. ENTRY *for* M.E. FRAME M123370

DATE	DAY TOTAL		TOTAL AIR TIME SINCE MFG.		NOTE: USE BOTH PAGES AND AS MANY LINES AS REQUIRED FOR COMPLETE
	HRS.	MIN.	HRS.	MIN.	
			12347		
Aug 9/09	2	56	1337	43	ONE ENGINE - NEED TOWANEY (AS) FOR PARTS
Aug 24/09	-	-	1337	43	WORK ON PER HRA WIA #309 - MAIN THREE REPAIRS. AS FOLLOWS: FRESH AIR VALVE/CONTROL. INSP. G/S FROM EXISTING/UNUSED LOADING. - ENGINE -
SEP. 23/19	-	-	1337	43	WORK CARRIED OUT BEGINNING JULY 3314, AND 3330, ALL FIRED IN THESE LOCS (PROP SENT FOR OIL + UH WIA #3081 - ENGINE/PROP. WORK FULL STRAINERS + FINISHED VALVE CORE + ORING + SERVICED
			1337	43	(CLOSED) (OCT. 31/14) WIA #3101 - INSTALLED NEW FUEL (H) RE-DICKED BEARINGS, R/H FUEL KIT AND HYD. FUELING PWS SENT TO AVIATION SHOP FOR NEW ONE FIRED WITH WIA #3101, DISASSEMBLED UH FUEL "SUMO" / HEADER TANK R/H CLEANED + O-RING/SEALS REPLACED 2
			1337	43	(CLOSED) (OCT. 31/19) WIA #3289 REPLACED UH MAIN TIRE/TUBE CHUNDER RECD. NEW O-RING + SEALS VALVE DISASSEMBLED/ NEW ORINGS + PUKED WITH NEW; NEW W/S PROCURED O.K. ° ELEVATORS/ RIDDER WERE PRE 9012 - (12-10) - AFTER CH A.D.S. (DISSED), SURFACE CORROSION ON TIP RIBS + STAMPED SURFACE CORROSION CLEANED REPAIRED WITH MSC. PARTS RECD. AIR HARDWARE; TRIM TAB PUSH PULK RECD. CLEANED/ RE-GRADED WITH PROPER GRADE, TRIM SYSTEM REBUILT TO SPEC. + SCREWS/ WASHERS (NOTE: A.D. JETS 315 (REV.1) WITH "NO RETORQUE" PL. COMPLIED WITH ON TIP RIB (PART I - CARRIED FORWARD REPEATING INSPECTION)

INSPECTION, REPAIRS & REPLACEMENT

ENTRIES: DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

DATE	TIME	SIGNATURE	LICENCE NUMBER
REPAIRS SIGNED ON W/O + COPIES PLACED IN TOOLBOXES FOR RECORD, WORK PLAN'S HEAT ADDITIONAL COOLING PIPES TO CONDENSER INTAKE TO IMPROVE PROX. BACK			
MAY 23 1970			
AND ENDING SEP. 30/70 PER HRA W/O's # 3881 3191 3289 RECORDS FOR ALL ENGINE/PROP. WORK, SET THEIR RESPECTIVE A.D. # 3006-18-15 AND SUPERVISED BY 3009-22-03.			
PID. OCS, REPLACED NOSE TIRE + BEARING RINGS, CORRUPT FUEL SYSTEM, SERVICED NOSE START WITH FUEL/REPLACED WITH NEW CHAIN DRIVE "INNUIT" JAWB ERTTS.			
FUELER SEAKERS (SK173-GI), REPLACED RTH MAIN TIRE + TUBE + SPRAY SERVICED WITH NEW SALES/ORINGS/CUP RINGS/STRAP			
REPLACED "SCARADER" UNDER FUELER VALVE WITH NEW GASKET 430 INTERNAL BUTTERFLY, NEW DATA CARD, + NEW DATA BASE (FORT)			
CLEANED + REASSEMBLED BRASS SLAVE CHANDLERS WITH NEW ORINGS + SENT FOR WEED REPAIR PLUS DRAIN ASSY. DISASSEMBLED			
EVA. STICK CAMPS SENT FOR WEED REPAIR + REPA. EVA. CHECKS WITH USED "NEAR NEW" AND REPACKED BEARINGS. RTH DOWNLOCK			
PINS RESET, + FIXED ALONG WITH COMD + TRAD SWITCH, PAIR ANGLE BACKUP RINGS INST. + BARREL BIRD, ONE ANTENNA ON BEAM RE			
"BUT NOT INSTALLED" OLD WLS HAS BEEN REPAIRED + STILL VIKING RETAINED TO CWL A.D. 3011-07-12 AND NOW SUPERCHARGED TO HORNETS "MUD" NEST REMOVED FROM EVU/ELEVATOR, CLEANED UP USING			
HINGE BRACKETS TREATED + PAINTED, CENTRAL ELEVATOR BECRANK UP! PAINTED + TREATED, ALL HINGE BEARINGS/BUSHINGS COINED +			
AND ELEVATORS/BECRANK/HINGES RESET WITH ALL NEW BEAD BURSTED/REPRIMED + PAINTED, TRIM SCREW JACKS DISASSEMBLED			
TRIM SCREWS/LINKS REINSTALLED WITH NEW HARDWARE, ELEVATORS SETTLED, ELEVATOR TIPS RESET WITH NEW CAP AND S			
- (3-10 MAY BE CWL IN THE FUTURE BY ATCC # 115W-1) - (QUERIED); PRIOR TO RUDDER RESET, SB 112-66, PARTS I AND II			
REPERITIVE INSPECTION AND PART II, MODIFICATION TO ELIMINATE			
CONTINUED NEXT PAGE - 10			

I.E. FRAME
MY23370

[illegible]CARRIED FORWARD

INSPECTION, REPAIRS & REPLACEMENT

ENTRIES: DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

DATE - ALL ROOFER HINGES

ENTRIES, DRAW A DIAGONAL LINE THROUGH ANY UNUSED LINES IN DATE AND TIME COLUMNS.

DATE	TIME	SIGNATURE	NUMBER
FROM PREVIOUS PAGE - ALL RUDDER HINGES / HINGE SPRING + IF READY, REWARD + RUDDER REINSTALLED WITH NEW HARDWARE. (WORKS STANDING ON WORK REPORTS FILED WITH WIO 3289). GEAR LIGHTS + CANNON LIGHT WIRING REPAIRED, GEAR SHIELD PAINTED SENT TO AIRCRAFT. FOR CALIBRATION / GEAR CHECK AND ANY CHANGES NEEDED / REPAIRED + PAINTED; FIRST MOUNT PLATE REINSTE WITH NEW HARDWARE; PILOT HEAT WARES, REPAIRED PILOTS; NEW THERMAL HEAT EXCH INSTALLED (FOR PILOT WITH WIO 3631) SYSTEM BLOWN OUT + LEAK CHECKED, ALF. STATIC SOURCE VALVE CORP. (FOR ONE FIVE WITH WIO); COMPLETED PILOT / STATIC LEAK "IN WHITES"; REPAIRED "B" NUT ON DARK BRASS DEVIATOR REPAIRED FUSE. CANNON PISTOLS AND STARTED REPAIRS TO G/HEAD DE FUSE. VENT DUCTING) / SIDE PANELS. (WIO CIRCLED END OF QUARTER SENT TO GARDIN FOR REPAIR / 8130-3 FILED WITH WIO); RVPMS BENCH TEST, ADJUSTMENT / INVENTIONS; COMPLETED REPAIRS / INSTALLED "TEMP." SERT COVERS ON FLYING STAYS; GARDIN ONLY"; FILED AIRCRAFT + LEAK CHECKED SYSTEM / REPAIRED AIRCRAFT BOX + INST. NEW G35 BATTERY SIN 602937871; AND TESTED; CHAL GEAR SALING; CHAL FUEL RUNNING ON GUARDING SYSTEM + VULCAN SYSTEM; CHAL TACH CHECK CANNON SALING (NEW CORP). AIRCRAFT IS CONSIDERED FLYABLE ANNUAL RE-CERT. IF THE FOLLOWING DEFECTS ARE ADDRESSED: OUT OF TOLERANCE (STILL FLYABLE) AID KIT (OR) FIRE EXTINGUISHER ("FLYABLE" UNDER PERMIT) OVERDUE BUT CAN BE FERRIED OUTSIDE CLASSES B, C, OR DESIS. D. STRICTS RANGE. "OVER" INSPECTIONS SIGNED ON WIO'S SIGNED IN JOURNEY LOG FOR CREW REF. - 46			

IE FRAME
M133570